

ISPR 2017 TU WIEN, Sept 2017



Structural Integrity of Complex Components by Means of the Strain Energy Density Approach

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ISPR 2017 13-15 September 2017, Vienna







Outline of Presentation



Introduction
Objectives
Theoretical Background
Testing
SED applied for fatigue assessment
Ongoing projects



















30000 students 40 faculties Nobel prize award 2014



- 1968 Lars Onsager, Chemistry 🥝 (graduate engineer from Norwegian Institute of Technology, NTH, in 1925)
- 1973 Ivar Giaever, Physics 🥝 (graduate engineer from Norwegian Institute of Technology, NTH, in 1952)
- 2014 Edvard Moser, Medicine or Physiology (professor of neuroscience, NTNU)
- 2014 May-Britt Moser, Medicine or Physiology (professor of neuroscience, NTNU)
- 2014 John O'Keefe, Medicine or Physiology 🥝 [31] (visiting researcher, NTNU 2015–)











Department of Mechanical Engineering



Stephen Hawking



Buzz Aldrin



May-Britt Moser



Emmanuelle Charpentier















Katharine Hayhoe

Atmospheric scientist



Sara Seager

Planetary scientist



Oliver Stone Director and writer



Structural Mechanics and Materials Research Group



Department of Mechanical Engineering

Staff in my group:

- **3** Academics
- 2 Research Fellows
- ~ 10 PhD students

Supported by workshop and admin staff

Research Directions:

Fatigue of metallic and non metallic materials Solid and Fracture Mechanics Structural Integrity of welding under fatigue loadings Material Science Full scale testing Structural Health Monitoring

Structural Mechanics and Materials Research Group



Facilities:

Welding facilities Mechanical testing capabilities (more than 80 MEuro) NanoLab

Lazzarin fatigue Lab (15 MEuro)

Collaboration:

Collaborative work with Norwegian, Australian, European, Canadian and South Korean Universities and Research Centres













OBJECTIVES



The present work investigates the multiaxial fatigue strength of sharp V-notched components made of titanium grade 5 alloy (Ti-6Al-4V).

Axi-symmetric notched specimens have been tested under combined tension and torsion fatigue loadings, both proportional and non-proportional, taking into account different nominal load ratios (R = -1 and R=0) and biaxiality ratios ($\lambda=\tau/\sigma=2$ and 0.6). All tested samples have a notch root radius about equal to 0.1 mm, a notch depth of 6 mm and an opening angle of 90 degrees.

Altogether, more than 250 fatigue results (19 S-N curves) are examined, first on the basis of nominal stress amplitudes referred to the net area and secondly by means of the strain energy density averaged (Lazzarin and Zambardi, 2001) over a control volume embracing the Vnotch tip. 15











Strain Energy Density



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Gdoutos EE (1990) Fracture Mechanics Criteria and Applications, Dodrecht: Kluwer Academic Publishers; 1990.

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VARIATION OF THE STRAIN ENERGY DENSITY AT THE NOTCH TIP FOR BLUNT NOTCHES



$$W^{(e)} = \frac{1}{2E} \left[\sigma_{rr}^{2} + \sigma_{\theta\theta}^{2} + \sigma_{zz}^{2} - 2\nu (\sigma_{rr} \sigma_{\theta\theta} + \sigma_{rr} \sigma_{zz} + \sigma_{\theta\theta} \sigma_{zz}) + 2(1+\nu)(\sigma_{r\theta}^{2} + \sigma_{rz}^{2} + \sigma_{\theta z}^{2}) \right]$$

$$\Delta W_{t} = c_{w} \cdot \frac{1}{2 \cdot E} \cdot \left(\Delta \sigma_{p,e1}^{2} + 2 \cdot (1+\nu) \cdot \Delta \tau_{p,e1}^{2} \right)$$

$$\Delta W_{t} = c_{w} \cdot \frac{1}{2 \cdot E} \cdot \left(k_{t\,net,ax}^{2} \cdot \Delta \sigma_{net}^{2} + 2 \cdot (1+\nu) \cdot k_{t\,net,tors}^{2} \cdot \Delta \tau_{net}^{2} \right)$$

$$= \begin{cases} \frac{1+R^{2}}{(1-R)^{2}} & \text{if } -1 \le R < 0 \\ 1 & \text{if } R = 0 \\ \frac{1-R^{2}}{(1-R)^{2}} & \text{if } 0 < R \le 1 \end{cases}$$









$$\Delta \overline{W} = c_{w} \left\{ \frac{e_{1}}{E} \left[\frac{\Delta K_{1}}{R_{0}^{1-\lambda_{1}}} \right]^{2} + \frac{e_{2}}{E} \left[\frac{\Delta K_{2}}{R_{0}^{1-\lambda_{2}}} \right]^{2} + \frac{e_{3}}{E} \left[\frac{\Delta K_{3}}{R_{0}^{1-\lambda_{3}}} \right]^{2} \right\}$$

Plane strain Ax

Axis-sym.

2lpha [rad]	γ [rad]	λ_1	λ_2	λ_3	e_1	e_2	e ₃
0	π	0.5000	0.5000	0.5000	0.13449	0.34139	0.41380
$\pi/12$	$23 \pi/24$	0.5002	0.5453	0.5217	0.13996	0.30588	0.39659
$\pi/6$	$11 \pi / 12$	0.5014	0.5982	0.5455	0.14485	0.27297	0.37929
$\pi/3$	$5\pi/6$	0.5122	0.7309	0.6000	0.15038	0.21530	0.34484
$\pi/2$	$3\pi/4$	0.5445	0.9085	0.6667	0.14623	0.16793	0.31034
$2\pi/3$	$2\pi/3$	0.6157	1.1489	0.7500	0.12964	0.12922	0.27587
3π/4	$5\pi/8$	0.6736	1.3021	0.8000	0.11721	0.11250	0.25863

$$Performance in the constraint of the characteristic decision of the characteristic decision$$

<u>Advantages</u>



- Permits consideration of the scale effect
- Permits consideration of the contribution of different Modes
- Permits consideration of the cycle nominal load ratio
- Overcomes the complex problem tied to the different NSIF units of measure in the case of crack initiation at the toe $(2\alpha=135^\circ)$ or root $(2\alpha=0^\circ)$
- SED can be evaluated with coarse meshes
- It directly takes into account the T-stress
- It directly includes three-dimensional effects
- Direct link with other local approaches (NSIFs, PS method, critical distances)
 - Not specific coefficient to be set up

<u>Tests</u>



Analysis of the bi-axial fatigue behaviour of notched specimens in Ti6Al4V (Gr. 5) alloy under combined tensile and torsion loading, both in-phase and out-ofphase, with particular reference to the evaluation of HCF Application of an energy-based approach to summarise

Application of an energy-based approach to summarise the fatigue data in presence of severe notches

- Tensile fatigue tests on notched specimens (R=-1 and R=0 f=5-20 Hz)
- Torsional fatigue tests on notched specimens (R=-1 and R=0, f=5-20 Hz)
- Combined tensile-torsional fatigue tests on notched specimens
- Analysis of the influence of load ratio on the multiaxial fatigue behaviour (R=0 and R=-1, f=5-20 Hz)
- Analysis of the influence of biaxiality ratio on the multiaxial fatigue behaviour (λ =0.6 and 2 , f=5-20 Hz)
- Analysis of the influence of the phase angle on the multiaxial fatigue behaviour ($\Phi=0$ and $\Phi=90^{\circ}$, f=5-20 Hz)

















$$Pertonector Construction Cons$$





$$\mathbf{R}_{1} = \left(\sqrt{2 \mathbf{e}_{1}} \times \frac{\Delta \mathbf{K}_{1\mathrm{A}}}{\Delta \sigma_{1\mathrm{A}}}\right)^{\frac{1}{1-\lambda_{1}}} \mathbf{R}_{3} = \left(\sqrt{\frac{\mathbf{e}_{3}}{1+\nu}} \times \frac{\Delta \mathbf{K}_{3\mathrm{A}}}{\Delta \tau_{3\mathrm{A}}}\right)^{\frac{1}{1-\lambda_{3}}}$$

For mode 1 loading: $\Delta K_{1A} = 452 \text{ MPa} \cdot \text{mm}^{0.455} \text{ ; } \Delta \sigma_{1A} = 950 \text{ MPa}$ For mode 3 loading: $\Delta K_{3A} = 1216 \text{ MPa} \cdot \text{mm}^{0.333} \text{ ; } \Delta \tau_{3A} = 776 \text{ MPa}$





$$R_1 = 0.051 \text{ mm}$$

 $R_3 = 0.837 \text{ mm}$

STRAIN ENERGY DENSITY AT THE NOTCH TIP MODE 1 + MODE 3

 \mathbf{R}_1

90°

 $\succ \Delta K_1$ and ΔK_3 are the NSIFs range under Mode I and Mode III:

 $\Delta \mathbf{K_1} = k_1 d^{1-\lambda_1} \Delta \sigma_{\text{nom}} = 2.260 \,\Delta \sigma_{\text{nom}} \quad (\text{MPa mm}^{0.445})$

 $\Delta K_3 = k_3 d^{1-\lambda_3} \Delta \tau_{nom} = 2.096 \Delta \tau_{nom}$ (MPa mm^{0.333})

➢ $e_1 = 0.146$ and $e_3 = 0.310$ depend on the notch opening angle and Poisson's ratio;

The control radii depend on the fatigue strength of unnotched specimens and on the stress range of the NSIFs values:

$$\mathbf{R_1} = \left(\sqrt{2\mathbf{e}_1} \cdot \frac{\Delta \mathbf{K}_{1\mathrm{A}}}{\Delta \sigma_{1\mathrm{A}}}\right)^{\frac{1}{1-\lambda_1}} = 0.051 \,\mathrm{mm} \,\,\mathbf{R_3} = \left(\sqrt{\frac{\mathbf{e}_3}{1+\nu}} \cdot \frac{\Delta \mathbf{K}_{3\mathrm{A}}}{\Delta \tau_{3\mathrm{A}}}\right)^{\frac{1}{1-\lambda_3}} = 0.837 \,\mathrm{mm}$$

















Coarse mesh









Coarse mesh







Coarse mesh



3D models	Number of FE	Degrees of freedom	W	K ₁	Δ%
	in the volume	(complete model)	Nmm/mm ³	[MPa mm ^{0.520}]	
1	1696	$8.6 \cdot 10^5$	0.07937	373.5	0
2	768	$4.6 \cdot 10^5$	0.07903	372.7	0.21
3	324	$2.5 \cdot 10^5$	0.07896	372.5	0.26
4	96	$1.7 \cdot 10^5$	0.07895	372.5	0.26
5	24	$4.5 \cdot 10^4$	0.07790	370.0	0.93
6	4	$1.1 \cdot 10^4$	0.07594	365.3	2.18



Coupled Fracture Modes Department of Mechanical Engineering 2,0 Crack front – Free surface Mid-plane→ 1,5 $\frac{K_{II}(z/h)}{}$ - Primary fracture mode Z KII 1,0 Poisson's ratio increasing $\frac{K_{III}(z/h)}{2}$ - Coupled fracture mode 0,5 00000000000 0.0 0,0 0,1 0,2 0,3 0,4 0,5 0 Free surface z/h Mechanism of generation of coupled mode: Poisson's Effect













Fig. 1 Si specimen with notch.



Fig. 2 Scanning electron microscopy image of the specimen with notch.

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Intermediate scale















Additive Materials





Additive Manufacturing (AM) materials under fatigue loading from full scale to nano level.

Examination of the state-of-the-art shows that fatigue assessment and quality assurance of additively manufactured components cannot be performed accurately due to a lack of bespoke methodologies allowing the specific microstructural features as well as the specific mechanical/cracking behaviour of additively manufactured materials to be modelled effectively.

Additive Materials





Topological complexity



Material/Hierarchical complexity



Functional complexity





Quality assurance

Additive Materials Interactions btw notches and defects

Department of Mechanical Engineering









Additive Materials Interactions btw notches and defects

Additive Materials Interactions btw notches and defects

EUROPEAN-EXTREMELY LARGE TELESCOPE (E-ELT) 1.200.000 weldments, 1.500.000 notches

Very proud!!!! (design of the gates and of the ships for the transportation 10.000 Km by sea) Cimolai has financed 5 PhD (500 KEuro) and employed 10 master students opening a research centre dedicated to advanced structural problems

AUTOMATIC FATIGUE CHECK OF WELDS AND NOTCHES

Department of Mechanical Engineering

Dedicated post-processor for massive fatigue check of weldings, notches and base material.

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- Nominal stress
- Modified nominal stress
- SED and N-SIF (shell element based)
- Hot-Spot stress
- □ IMPLEMENTED STANDARDS:
 - Eurocode 3 Design of steel structures: Fatigue
 - IIW Recommendations for fatigue design of welded joints, Hobbacher
 - EN 13445 DNV-GL (Hot Spot user defined)

AUTOMATIC FATIGUE CHECK OF WELDS AND NOTCHES

Additive Materials

ESIS Technical committee 15 (Chief) ESFRI HORIZON 2020 PROPOSAL (180 ME, 13 **PARTNERS)**

The fatigue behaviour of Ti6Al4V under multi-axial loading and in presence of notches was investigated

- The nominal load ratio was found to have a great influence on the fatigue strength while the influence of the load phase angle seems to be limited
- The re-analysis of the fatigue data in terms of the strain energy density range at the notch tip allowed us to summarise all the multiaxial data on notched specimens in a single fatigue scatter band

The proposed approach results therefore a very useful tool for the fatigue analysis of components subjected to multiaxial fatigue loading in the presence of severe notches ⁶⁰

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You are also welcome in Norway!!!

ISPR 2017 13-15 September 2017, Vienna

MANY THANKS FOR THE KIND INVITATION

HONORED AND GRATEFUL TO BE HERE

it was better to meet you at least once then never have met you

'My favorite things in life don't cost any money. It's really clear that the most precious resource we all have is time'